

Investigating Altitude and Velocity

| Team Members | |
|--------------|--|
| | |
| | |
| | |

Mission: Your team will determine the average velocity of the Wildcat 20mm model rocket with a 3 fin configuration and report the results to the class.

Summary:

- Your team will launch your rockets to determine each rocket's apogee using a sextant and determine the mode (The most frequent value in the data set) and the median (The middle value that separates the higher half from the lower half of the data set) averages.
- You will also determine the average velocity of the model rockets.

Materials:

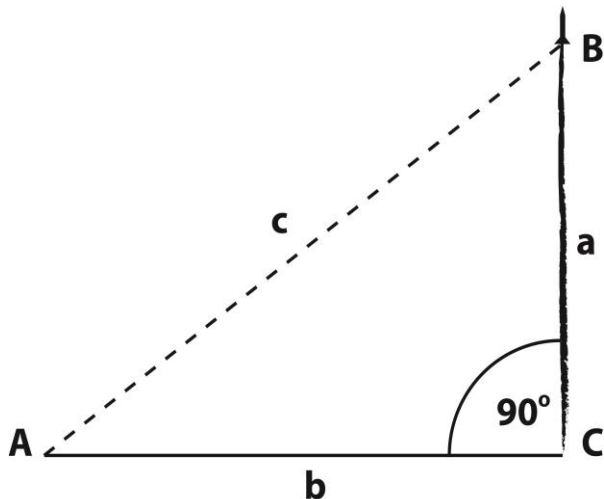
1. A completed model rocket for each member. All rockets must have the same weight in grams.
2. 2 trackers with Estes Altitracs at two separate locations
3. 3 timers with stopwatches
4. Class launch equipment.

Sequence of Events:

1. Each team member will build a model rocket to the teacher's specifications.
2. Adjust their weights in grams to be equal. Determine how to do this as a team.
3. Request a launch date from the teacher.
4. On prep day, prep each rocket using an Engineer Checklist for each rocket. Load each rocket with an A8-3 rocket motor.
5. Your team will operate the control center and record any collected data.
6. Launch each of your rockets on the same day, if possible. Gather required information from trackers, timers, and meteorologists and enter the data onto the flight log
7. After all of your rockets have been launched, fill out the Altitude and Velocity Worksheet and complete all calculations.
8. Give a report to the class on your findings.
9. Submit all paperwork to the teacher.

Calculating Altitude with a Sextant

To determine the apogee of a rocket's flight, we collect two types of data: distances and angles. This data is used to create a triangle with is a model of the lines that would join the tracker and the rocket, the rocket and a point directly below the ground, and the point on the ground and the tracker.



In this diagram, A represents the tracking station, B is the rocket at maximum altitude, and C is the launch pad. The angle formed by the lines at C is then a right angle or 90 degrees. Since there are 180 degrees in the angles of a triangle, if we know angle A, we can find angle B, since $B = 180 \text{ degrees} - (A + C)$, or $B = 90 \text{ degrees} - A$. (In trigonometry, a capital letter represents an angle, a small letter represents a side. The small letter "a" will always be used to represent the side opposite angle A, "b" the side opposite B, etc. Two capital letters together represent a distance. Thus BC represents the distance from angle B to angle C, or side "a".

If the rocket flight is vertical, we can call C a right angle (90 degrees). In that case, B is equal to $90 \text{ degrees} - A$. So we must measure angle A during the flight of the rocket using a sextant. A sextant is a measuring device that gives us the angle of an object in degrees from the ground. Students use a sextant called an Estes Altitrak, which is designed to track model rockets.

We need to know the distance of the tracker using the sextant, which we call the tracking station, from the launch pad, which is where the rocket will begin its flight. The recommended distance is twice the distance of how high we expect the rocket to fly, but if that is not possible, it should be as far away as possible for accuracy. We use a measuring wheel to measure that distance. The distance can be in feet or meters.

When the rocket launches, the tracker follows the rocket up with the sextant by aiming at the rocket's nose. At the highest point of the flight, the tracker releases the trigger of the AltiTrak which locks the measuring arm over a number that represents angle A in degrees. By using the Table of Tangents, the tangent of angle A can be determined.

To find the distance from C (the launch pad) to B (the highest point of the rocket's flight), we take the tangent of angle A and multiply it by the distance from the tracker to the launch pad (side AC).

$$\text{Tan } A \times AC = BC$$

To convert meters to feet - multiply by 3.2808, To convert feet to meters - divide by 3.2808

Tangent Table

| Degree | Tangent | Degree | Tangent | Degree | Tangent | Degree | Tangent |
|--------|---------|--------|---------|--------|---------|--------|---------|
| 1 | 0.0175 | 21 | 0.3838 | 41 | 0.8692 | 61 | 1.8037 |
| 2 | 0.0349 | 22 | 0.4040 | 42 | 0.9003 | 62 | 1.8804 |
| 3 | 0.0524 | 23 | 0.4244 | 43 | 0.9324 | 63 | 1.9622 |
| 4 | 0.0699 | 24 | 0.4452 | 44 | 0.9656 | 64 | 2.0499 |
| 5 | 0.0875 | 25 | 0.4663 | 45 | 1.0000 | 65 | 2.1440 |
| 6 | 0.1051 | 26 | 0.4877 | 46 | 1.0354 | 66 | 2.2455 |
| 7 | 0.1228 | 27 | 0.5095 | 47 | 1.0722 | 67 | 2.3553 |
| 8 | 0.1405 | 28 | 0.5317 | 48 | 1.1105 | 68 | 2.4745 |
| 9 | 0.1584 | 29 | 0.5543 | 49 | 1.1502 | 69 | 2.6044 |
| 10 | 0.1763 | 30 | 0.5773 | 50 | 1.1916 | 70 | 2.7467 |
| 11 | 0.1944 | 31 | 0.6008 | 51 | 1.2347 | 71 | 2.9033 |
| 12 | 0.2125 | 32 | 0.6248 | 52 | 1.2798 | 72 | 3.0767 |
| 13 | 0.2309 | 33 | 0.6493 | 53 | 1.3269 | 73 | 3.2698 |
| 14 | 0.2493 | 34 | 0.6744 | 54 | 1.3762 | 74 | 3.4862 |
| 15 | 0.2679 | 35 | 0.7001 | 55 | 1.4279 | 75 | 3.7306 |
| 16 | 0.2867 | 36 | 0.7265 | 56 | 1.4823 | 76 | 4.0091 |
| 17 | 0.3057 | 37 | 0.7535 | 57 | 1.5396 | 77 | 4.3295 |
| 18 | 0.3249 | 38 | 0.7812 | 58 | 1.6001 | 78 | 4.7023 |
| 19 | 0.3443 | 39 | 0.8097 | 59 | 1.6640 | 79 | 5.1418 |
| 20 | 0.3639 | 40 | 0.8390 | 60 | 1.7317 | 80 | 5.6679 |

ID06: Investigating Average Velocity



**Distance/Time
= Average Velocity**

$$d/t = v$$

$$60\text{m}/3\text{s} = 20\text{mps}$$

Model rockets travel at different rates of speed throughout the flight. During the first moments of flight, the rocket is accelerating. As it consumes its fuel, the rocket becomes lighter and speeds up until it reaches what is known as maximum velocity, or the fastest speed that it will go and now longer accelerate. Once the fuel is consumed, the rocket continues to fly because of the stored, or kinetic energy. This is known as the coast phase as the rocket is using up its kinetic energy and begins to decelerate, or slow down. The flight ends when the rocket deploys its recovery system. We know when this event occurs because we hear a popping sound as the deployment charge ignites and pushes out the recovery system.

In order to calculate the average velocity, we can time the flight from launch to apogee with a stopwatch. Once the trackers report the altitude of the flight, we can take that number and divide it by the time of flight from launch to apogee. The result is the average velocity in either feet per second (fps) or meters per second (mps).

ENGINEER CHECKLIST

ROCKET SERIAL # _____

BUILDER: _____

PRE-FLIGHT SAFETY CHECK

| GO | NO GO | |
|----|-------|--|
| | | All glue and paint on model is completely dry |
| | | Model is complete and all parts are present |
| | | Nose cone fits properly and is not tight |
| | | Nose cone is securely attached to the airframe |
| | | Shock cord is secure |
| | | Airframe is straight with no bends or warps |
| | | Fins are present, securely attached and properly aligned |
| | | Fins are undamaged |
| | | Launch lug is securely attached to the airframe |
| | | Motor mount is secure and operational |
| | | ROCKET IS READY FOR FLIGHT! |

PRE-FLIGHT PREPARATION

| | | |
|--|--|------------------------------------|
| | | ROCKET WEIGHT EMPTY: _____ grams |
| | | Wadding installed |
| | | Recovery system installed |
| | | Rocket motor nomenclature: _____ |
| | | Rocket motor undamaged |
| | | Rocket motor installed |
| | | Igniter and igniter plug installed |
| | | Payload description: _____ |
| | | Payload installed |
| | | ROCKET WEIGHT LOADED: _____ grams |

POST-FLIGHT INSPECTION

| | | |
|--|--|--|
| | | Rocket successfully recovered |
| | | Rocket nose, airframe, and fins are intact and undamaged |
| | | Recovery system is reusable |
| | | ROCKET POST-FLIGHT WEIGHT : _____ grams (Including engine casing & recovery system) |

LAUNCH LOG

Rocket Name: _____

Serial # _____

Builder: _____

| LAUNCH INFORMATION | FLIGHT DATA | |
|------------------------|-----------------------------|-------------------------------|
| Date: | Liftoff | Recovery |
| Launch Time: | Successful: | Recovery System Deployment |
| Location: | <i>Misfire</i> | <i>Stage 1</i> |
| Launch Pad Elevation: | Stage 1: | Before Apogee: |
| | Stage 2: | At Apogee: |
| | | During Descent: |
| ROCKET DATA | Pitch & Roll | |
| Fin Design: | <i>Thrust Phase</i> | Partial Deployment: |
| Fin # | No Pitch/Roll: | Failed to Deploy: |
| Engine | Pitched: | <i>Stage 2</i> |
| Stage 1: | Rolled: | Before Apogee: |
| Stage 2: | Tumbled: | At Apogee: |
| | Weathercock: | During Descent: |
| Recovery System | <i>Coast Phase</i> | Partial Deployment: |
| <i>Stage 1:</i> | Straight Trajectory: | Failed to Deploy: |
| Parachute - | Weathercock: | |
| Diameter: | Tumbled: | Recovery System Performance |
| Spill Hole Diameter: | | <i>Stage 1</i> |
| Streamer - | ALTITUDE | Stable Descent: |
| Size: | <i>Tracking Station</i> | Oscillation: |
| Material: | Track. 1 Distance from pad: | Spinning: |
| <i>Stage 2:</i> | Track.2 Distance from pad: | <i>Stage 2</i> |
| Parachute - | Track.3 Distance from pad: | Stable Descent: |
| Diameter: | Tracker 1 Degrees: | Oscillation: |
| Spill Hole Diameter: | Tracker 2 Degrees: | Spinning: |
| Streamer - | Tracker 3 Degrees: | |
| Size: | | Landing |
| Material: | <i>Marker Streamer</i> | Soft: |
| | Timer 1: | Hard: |
| Mass | Timer 2: | Crash: |
| Empty: | | Distance from Pad: |
| Loaded: | <i>Electronic Altimeter</i> | Direction from Pad: |
| Post: | Reading: | |
| | FLIGHT TIMES | Post Flight Inspection |
| METEOROLOGY | <i>To Apogee</i> | <i>Damage</i> |
| Temperature: | Timer 1: | Nose: |
| Humidity: | Timer 2: | Airframe: |
| Barometer: | <i>Apogee to Landing</i> | Fins: |
| Wind Speed: | Timer 1: | Shock Cord: |
| Wind Direction: | Timer 2: | Recovery System: |
| Conditions: | <i>Total Time of Flight</i> | Can be reflown? |
| Cloud Type: | Timer 1: | |
| | Timer 2: | |

AVERAGE ALTITUDE WORKSHEET

Use the raw data collected on the flight log of each rocket to fill out the form

Rocket 1

| | |
|-------------------------------|-------------------------------|
| Apogee in Degrees Tracker 1 | Apogee in Degrees Tracker 2 |
| Tangent A | Tangent A |
| Distance from Launch Pad (AC) | Distance from Launch Pad (AC) |
| Tan A x AC = BC | Tan A x AC = BC (altitude) |
| Average Altitude: | |

Rocket 2

| | |
|-------------------------------|-------------------------------|
| Apogee in Degrees Tracker 1 | Apogee in Degrees Tracker 2 |
| Tangent A | Tangent A |
| Distance from Launch Pad (AC) | Distance from Launch Pad (AC) |
| Tan A x AC = BC | Tan A x AC = BC (altitude) |
| Average Altitude: | |

Rocket 3

| | |
|-------------------------------|-------------------------------|
| Apogee in Degrees Tracker 1 | Apogee in Degrees Tracker 2 |
| Tangent A | Tangent A |
| Distance from Launch Pad (AC) | Distance from Launch Pad (AC) |
| Tan A x AC = BC | Tan A x AC = BC (altitude) |
| Average Altitude: | |

Rocket 4

| | |
|-------------------------------|-------------------------------|
| Apogee in Degrees Tracker 1 | Apogee in Degrees Tracker 2 |
| Tangent A | Tangent A |
| Distance from Launch Pad (AC) | Distance from Launch Pad (AC) |
| Tan A x AC = BC | Tan A x AC = BC (altitude) |
| Average Altitude: | |

Rocket 5

| | |
|-------------------------------|-------------------------------|
| Apogee in Degrees Tracker 1 | Apogee in Degrees Tracker 2 |
| Tangent A | Tangent A |
| Distance from Launch Pad (AC) | Distance from Launch Pad (AC) |
| Tan A x AC = BC | Tan A x AC = BC (altitude) |
| Average Altitude: | |

Rocket 6

| | |
|-------------------------------|-------------------------------|
| Apogee in Degrees Tracker 1 | Apogee in Degrees Tracker 2 |
| Tangent A | Tangent A |
| Distance from Launch Pad (AC) | Distance from Launch Pad (AC) |
| Tan A x AC = BC | Tan A x AC = BC (altitude) |
| Average Altitude: | |

Results

Mode:

Median:

AVERAGE VELOCITY WORKSHEET

Use the raw data collected on the flight log of each rocket to fill out the form

Rocket 1

| | | | |
|---------------------------------|--|----------------------------------|--|
| Launch to Apogee Timer 1: | | Time to Apogee Timer 2: | |
| Average Ascent Velocity: | | Average Ascent Velocity: | |
| Apogee to Ground Timer 3: | | Apogee to Ground Timer 4: | |
| Average Descent Velocity: | | Average Descent Velocity: | |
| Average Ascent Velocity: | | Average Descent Velocity: | |

Rocket 2

| | | | |
|---------------------------------|--|----------------------------------|--|
| Launch to Apogee Timer 1: | | Time to Apogee Timer 2: | |
| Average Ascent Velocity: | | Average Ascent Velocity: | |
| Apogee to Ground Timer 3: | | Apogee to Ground Timer 4: | |
| Average Descent Velocity: | | Average Descent Velocity: | |
| Average Ascent Velocity: | | Average Descent Velocity: | |

Rocket 3

| | | | |
|---------------------------------|--|----------------------------------|--|
| Launch to Apogee Timer 1: | | Time to Apogee Timer 2: | |
| Average Ascent Velocity: | | Average Ascent Velocity: | |
| Apogee to Ground Timer 3: | | Apogee to Ground Timer 4: | |
| Average Descent Velocity: | | Average Descent Velocity: | |
| Average Ascent Velocity: | | Average Descent Velocity: | |

Rocket 4

| | | | |
|---------------------------------|--|----------------------------------|--|
| Launch to Apogee Timer 1: | | Time to Apogee Timer 2: | |
| Average Ascent Velocity: | | Average Ascent Velocity: | |
| Apogee to Ground Timer 3: | | Apogee to Ground Timer 4: | |
| Average Descent Velocity: | | Average Descent Velocity: | |
| Average Ascent Velocity: | | Average Descent Velocity: | |

Rocket 5

| | | | |
|---------------------------------|--|----------------------------------|--|
| Launch to Apogee Timer 1: | | Time to Apogee Timer 2: | |
| Average Ascent Velocity: | | Average Ascent Velocity: | |
| Apogee to Ground Timer 3: | | Apogee to Ground Timer 4: | |
| Average Descent Velocity: | | Average Descent Velocity: | |
| Average Ascent Velocity: | | Average Descent Velocity: | |

Rocket 6

| | | | |
|---------------------------------|--|----------------------------------|--|
| Launch to Apogee Timer 1: | | Time to Apogee Timer 2: | |
| Average Ascent Velocity: | | Average Ascent Velocity: | |
| Apogee to Ground Timer 3: | | Apogee to Ground Timer 4: | |
| Average Descent Velocity: | | Average Descent Velocity: | |
| Average Ascent Velocity: | | Average Descent Velocity: | |

Results

Mode:

Median: